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Department history and professional profile

The Division of Aerospace and Advanced Control (DAAC) was founded in 1997, and it consists of several departments which cover R&D and special small-scale prototype and serial production. The R&D aims at internal in-house development and made-to-order consumer development projects. The department has experienced professionals in software, electronic hardware, mechanical design, modelling and simulation, and testing development activities with respect to development of state-of-the-art airborne, automotive and industrial, especially for avionics, control systems, power electronics, intelligent & embedded systems. DAAC has a great experience in arranging full-prototype/serial production with respect to ISO 9001 Standards, EASA and FAA Standards. DAAC has been successful as a member of EU Research Development Projects in Aerospace (CESAR, SCARLETT, ACTUATION 2015, ESPOSA, Clean Sky) as well as in the National Research Projects. Lots of our activities have been realized in cooperation with the Czech universities.



DIVISION OF AEROSPACE AND ADVANCED CONTROL RELIABLE PARTNER IN THE EUROPEAN AVIATION SUPPLY CHAIN

Division of Aerospace and Advanced Control

Main activities:

- HW design and development
- SW design and development
- Modelling and simulation
- Testing
- Research & technology development Customer technical support

Certification:

- LRQA ISO EN 9001 and ISO / IEC 27 001 and ISO / IEC 20 000-1
- POA certificate issued by EASA
- National "DOA" issued by CAA

References:

- Gold medal for SAM at the 43rd Brno international engineering fair
- Gold medal for NETZ at the 45th Brno
- customer offer at AERO, Prague
- of the Czech Republic

- Software development and testing in accordance with:
- ANSI C standards Misra C development tools for DSP, MCU, CPU
- RTCA/DO 178 B/C
- IEEE standards software engineering
- Rapid control prototyping (RCP) and code generation Cantata C / C++

Hardware development in accordance with:

- RTCA/DO-160G
- RTCA/DO-254

Development tools:

- Electronic circuit and PCB design, simulation tools (Altium)
- · MatLab, Simulink and power simulation tools
- National Instruments NI
- dSpace
- Solid Edge
- Prototype / serial production / service in accordance with:
- ISO EN 9001 and EASA Part 21, subpart G









Development

(MATLAB, Simulink)

Production

Special serial production Small scale production Prototypes Production support (testers, simluators, etc.) Service and customer support Full production documentation ISO EN 9001 and EASA, part 21 subpart G approval

Application

Engine control units Electric power distribution boxes Modular control systems System of aviation modules SAM Power converters EHA / EMA actuators Fuel and hydraulic pump control units Embedded and mechatronic systems



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• NATO AQAP 2110



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- international engineering fair
- Grand prix prize for the most complex

Membership:



Association of Research Organizations (AVO)

Research and technology development Model based design, simulation Automated code generation Rapid control prototyping HW development in accordance with DO 254 HW testing in accordance with with DO 160 G SW development in accordance with DO 178B/C ANSI C and Misra C standards

References

Control units and power converters for jet & turbo-prop & turbo-shaft engines

Description:



The control and power system integrates power converter and is designed for the turbine engines - PBS TP100 and PBS TS100 with max. power up to 180 kW and max. thrust up to 1,3 kN for PBS TJ100 engine. HW design versatility allows connection and control of turboprop and turboshaft engines. A part of the control unit system is power electronics for starter generator enabling fully controlled start of the engine.

Application:

PBS Velká Bíteš, a.s. - engine PBS TJ100, PBS TP100, PBS TS100



Description:

Electric power distribution box (EPDB), as a part of electrical power system (EPS), is a set of two distribution boxes inteded for twin engine small aircafts of CS23 category with 28VDC dual-channel electrical system with two batteries and two starter-generators. The main function is the reliable distribution of the electrical energy from airborne systems to airborne loads with respect to the pilot's requests, the equipment of the aircraft and flight phases. EPDB also provides monitoring and protection of the EPS. The EPDB device is controlled by the switches from the cockpit and is implemented in the EV-55 aircraft.

Control system for aircraft engine (E018)

Description:



E018 is dual-channel engine control unit used to control engine D-18T series 3M/3A. The control unit is used to control the engine in all regimes and monitor its state and function. This unit is part of the automatic engine control system, provides the function of generating electrical control commands, engine protection and ensuring the required engine operating conditions. It is designed for a wide family of cargo and civil aircrafts. The first flight tests will be carried out on the Antonov An-124 aircraft.

Application:

Application:

Product line Al450 engines



Aircraft turboprop engine control system

Description:

Control system for aircraft engines utilizes the most modern design and production technologies. The control system was developed in cooperation with GE Aviation Czech company. There were developed prototypes for the purpose of performance of development verification tests, including testing of prototypes directly mounted on the engine at the GE Aviation Czech engine testing facility.

Control unit for AI-450S-2 and AI-800S engines

Description:

ECU (E452) is a complex aircraft engine control system which consists of:

- two ECU modules
- one EMM module
- one VMM module
- · one connector module

Two-channel modular control system intened for AI450-S2 and AI-800S and BE2/BE2+ engines is assembled of two ECU0 control units, one EMM engine monitoring unit, one VMM vibromodule unit and connector CON0 unit. Each of these modules represents an independent functional unit the unit is designed as a distributed system. If one of the modules fails, it is easy to exchange it and the other modules functionality is not affected.

ECU (E452) ensures also communication with onboard systems through I/O, Arinc 429 protocol and communication with service system through ethernet interface.

Power converters for aerospace applications

Description:

Power converters for aerospace applications are designed to convert AC voltage from engine generators and to provide onboard 28 V DC power supply in accordance with MIL 704/EN 2282 standard. PCTP units are designed in modular concept to supply power from 1 to 9 kW. PCTP units are designed to provide power conversion, overcurrent and undervoltage protection, limitation of the supplied power, monitoring and communication with superior control system via standard communication databus

LDIS Aerospace and Advanced Control

Application:

PBS Velká Bíteš, a.s. – TP100 (turbo-prop) and TS100 (turbo-shaft) engines



Fuel pump control units

Description:

The main function of the control unit of the transport fuel pump is to regulate the speed of the BLDC engine in order to achieve a constant value of the engine speed and thus the pump, which transports fuel from the tank to the fuel system of the aircraft. The unit was developed for the training aircraft L39-NG, which is manufactured by company AERO Vodochody AEROSPACE a.s.

Mathematical modeling and simulations

Description:



Electric power distribution box



Application:

Evektor, s.r.o. - Aircraft EV55





Application:



Application:

Power converters for aerospace applications Fuel pump control units Control units and power converters for jet & turbo-prop & turbo-shaft engines

